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NUAA

Learning-based Control for Deployment and Retrieval of a Spinning Tethered Satellite Formation System

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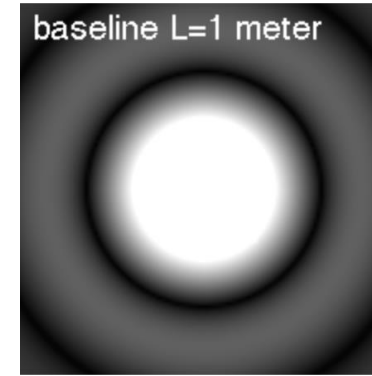
Nanjing University of Aeronautics and Astronautics

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- 3** Controller Design
- 4** Numerical Cases
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1 Backgrounds



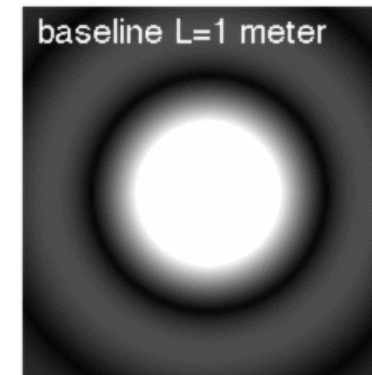
Space telescope with single aperture



Space exploration



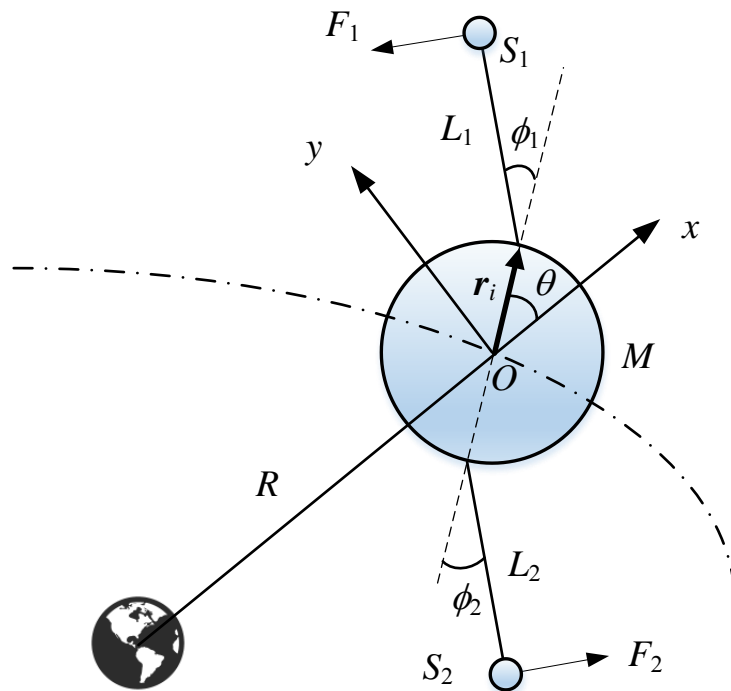
Tethered Space telescope with synthetic aperture



Focused Problems

- Attitude motions of tethers would be significantly affected due to deployment and retrieval of tethers. **Tangling motions** between the tethers and satellites may be generated and thereby destroy the system stability and safety.
- The system may have **asymmetric configurations** at initial due to perturbations, which is not good for keeping stability and should be considered.
- The system usually has **complex practical constraints**, such as the tether tension should not be too large to prevent cut-off of tethers. The control force should be kept in an acceptable range due to the limited maneuver ability of satellites, and so on.
- It is tough to compute control input online when the underlying control problem and the designed control law are complicated since the **onboard computing power of the satellites is limited**.

2 Dynamic model



Spinning tethered satellite formation (STSF)

$$\mathbf{R}_M = R\mathbf{e}_x, \quad \mathbf{R}_i = R\mathbf{e}_x + \mathbf{r}_i$$

$$\mathbf{v}_M = \dot{\mathbf{R}}_M = \Omega R\mathbf{e}_y, \quad \mathbf{v}_i = \dot{\mathbf{R}}_i = \Omega R\mathbf{e}_y + \dot{\mathbf{r}}_i$$



$$T = \frac{1}{2} m_M \mathbf{v}_M \cdot \mathbf{v}_M + \frac{1}{2} \sum_{i=1}^2 m_i \mathbf{v}_i \cdot \mathbf{v}_i + \frac{1}{2} J(\dot{\theta} + \omega)^2$$

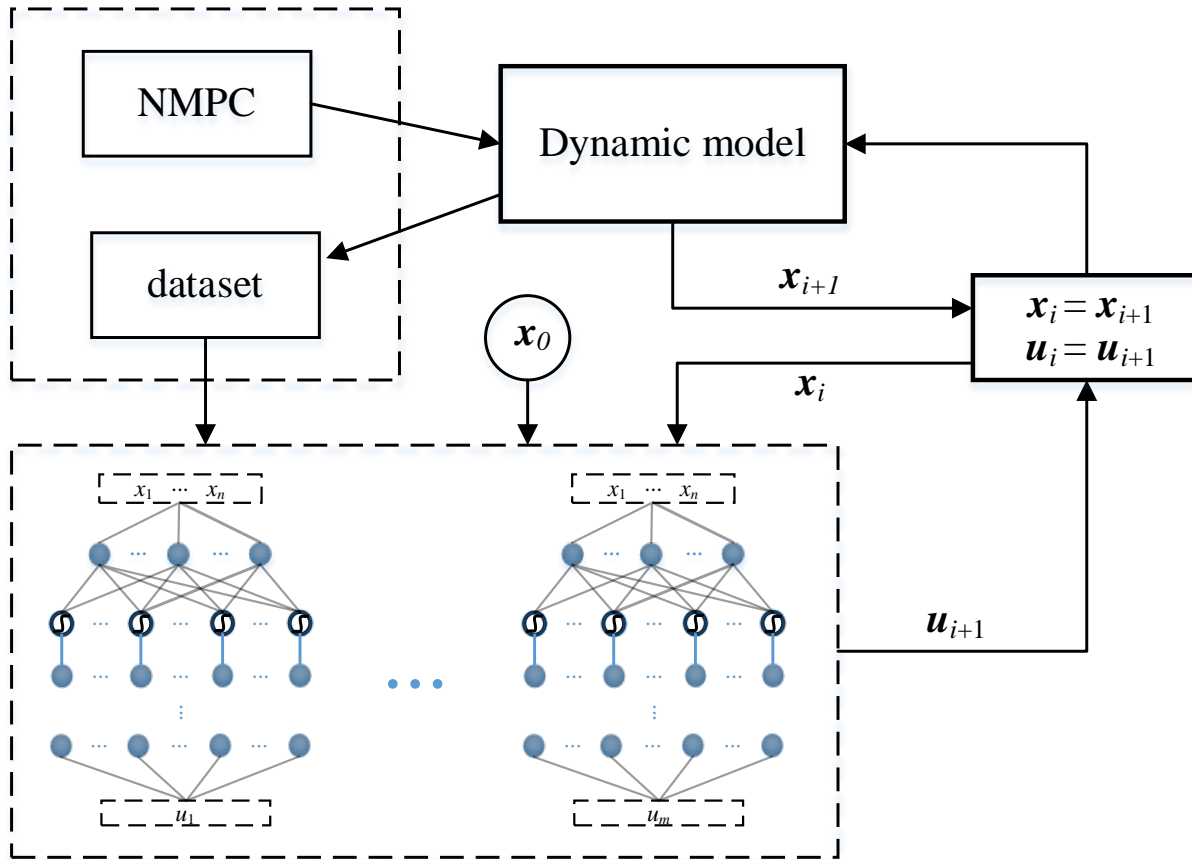
$$U = -\frac{\mu m_M}{|\mathbf{R}_M|} - \sum_{i=1}^2 \frac{\mu m_i}{|\mathbf{R}_i|} = -\frac{\mu m_M}{R} - \sum_{i=1}^2 \frac{\mu m_i}{|\mathbf{R}_M + \mathbf{r}_i|}$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_j} \right) - \left(\frac{\partial L}{\partial q_j} \right) = Q_{q_j}, \quad j = 1, 2, 3, 4, 5$$



$$\mathbf{M}\ddot{\mathbf{q}} + \mathbf{C}\dot{\mathbf{q}} + \mathbf{K} = \mathbf{F}$$

● Design concept



Flowchart of learning-based controller

- Obtain dataset using NMPC;
- Construct DNN based on dataset;
- Input initial state x_0 to DNN to obtain control input quickly;
- Achieve the control goal.

● NMPC controller

➤ Optimal control model

State vector $\mathbf{x} = (x_1, x_2, x_3, x_4, x_5, x_6, x_7, x_8, x_9, x_{10})^T$
 $= (\theta, \phi_1, \phi_2, l_1, l_2, \dot{\theta}, \dot{\phi}_1, \dot{\phi}_2, \dot{l}_1, \dot{l}_2)^T$

Control vector $\mathbf{u} = (u_1, u_2, u_3, u_4)^T = (T_1, T_2, F_1, F_2)^T$

State-space form $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, t)$

Moving horizon Prediction horizon $T_p = N\sigma_t$

find $[t_{k+1}, t_{k+1} + T_p] \ni t \mapsto \{\mathbf{x}(t), \mathbf{u}(t)\}$

to minimize $f_J = \int_{t_{k+1}}^{t_{k+1} + T_p} f_F(\mathbf{x}, \mathbf{u}, t) dt$

subject to $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, t)$

$\mathbf{x}(t_{k+1}) = \hat{\mathbf{x}}_{k+1}$

$u_L \leq u_1, u_2, u_3, u_4 \leq u_U$

➤ Solving process

kth computation $[t_k, t_{k+1}] \longleftrightarrow t_{k+1} = t_k + \sigma_t$
 Sampling interval

NLP Problem

$$f_J \approx \sum_{i=1}^N \sum_{j=1}^{N_r} W_j^{(i)} f_F(\mathbf{x}_j^{(i)}, \mathbf{u}^{(i)})$$

$$\sum_{j=1}^{N_r+1} D_{ni} \mathbf{x}_j^{(i)} - \mathbf{f}(\mathbf{x}_n^{(i)}, \mathbf{u}^{(i)}, \tau_{i,n}) = 0$$

$$\mathbf{x}_1^{(k+1)} = \hat{\mathbf{x}}_{k+1}$$

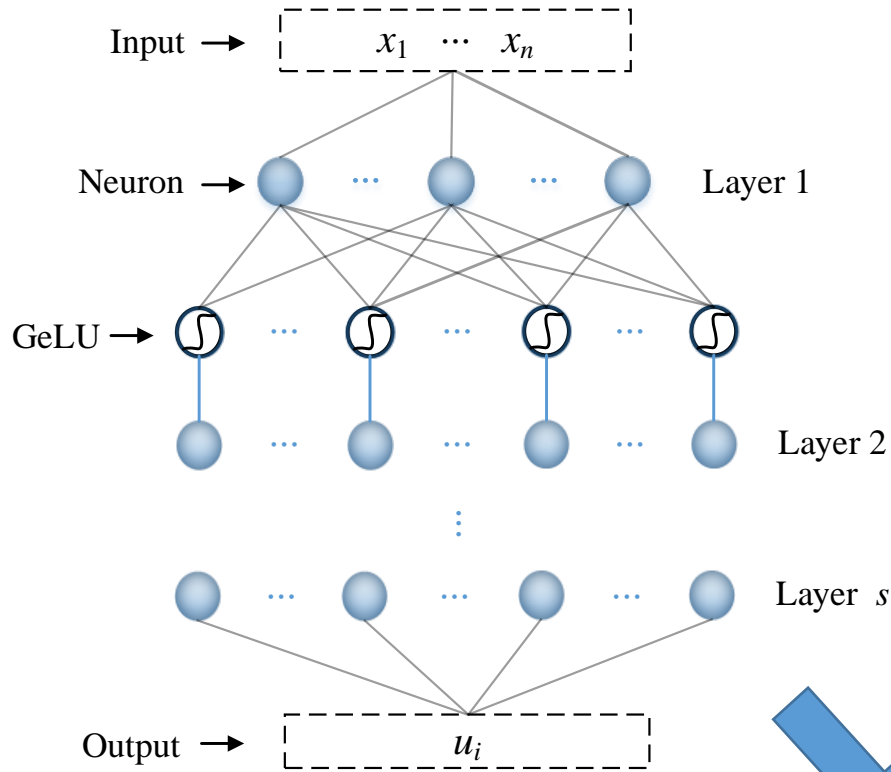
$$u_L \leq (u_{i,j})_1, (u_{i,j})_2, (u_{i,j})_3, (u_{i,j})_4, \leq u_U$$

IPOPT Solver

NMPC law $\longrightarrow [t_{k+1}, t_{k+2}]$

• DNN method

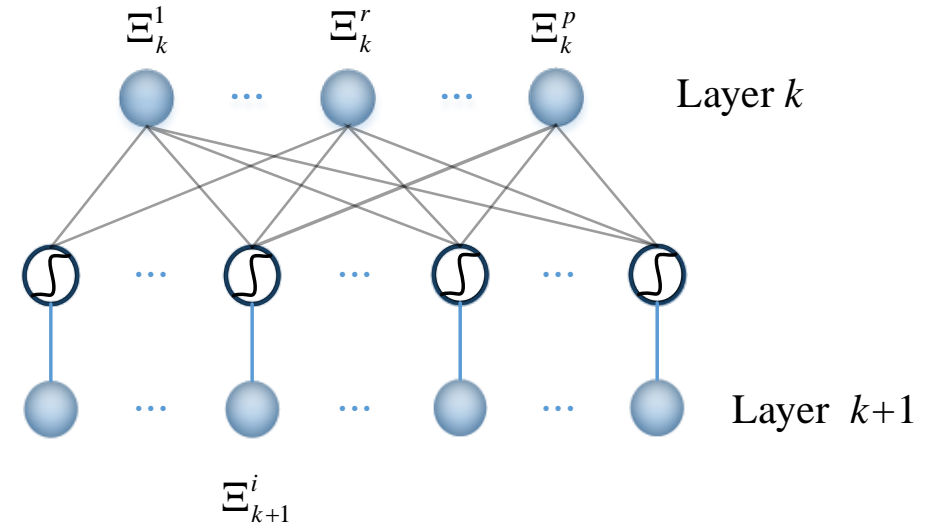
➤ Deep neural network



An s -layer DNN



$$u_i = f_{DNN}(\mathbf{x})$$



Relationship between two adjacent layers

Output $\Xi_{k+1}^i = f_{ac}^k(\mathcal{R}_k^{i,1}\Xi_k^1 + \dots + \mathcal{R}_k^{i,r}\Xi_k^r + \dots + \mathcal{R}_k^{i,p}\Xi_k^p + B_k^i)$

Weight \uparrow Bias \uparrow

$$\text{GeLU}(x) \approx 0.5x[1 + \tanh(\sqrt{\frac{2}{\pi}}(x + 0.0047715x^3))]$$

Parameters setting

Table. 1 Physical parameters

Physical parameters	Values
Mass of main satellite	1×10^4 kg
Mass of sub-satellite	50 kg
Radius of main satellite	5 m
Orbital radius	6878 km

Table. 2 Control parameters

Control parameters	Values
Number of sampling intervals	20
Number of collocation points	4
Tether tension	[0.1, 10] N
Control force	[-0.5, 0.5] N

Table. 3 DNN parameters

DNN parameters	Values
Number of hidden layers	10
Number of neurons in hidden layers	256, 128, 64, 64, 32, 32, 16, 16, 8, 5
Activation function of hidden layers	GeLU
Activation function of output layer	linear
Loss function	MSE
Optimizer	Adam
Batch size	256
Number of epochs	300
k -fold cross-validation method	$k = 6$

$$f_{FD} = 100(u_3^2 + u_4^2) + 500(x_2^2 + x_3^2) + 0.01[(x_4 - 100)^2 + (x_5 - 100)^2] + 100(x_9^2 + x_{10}^2)$$

$$f_{FR} = 100(u_3^2 + u_4^2) + 500(x_2^2 + x_3^2) + 0.01[(x_4 - 10)^2 + (x_5 - 10)^2] + 500(x_9^2 + x_{10}^2)$$

4 Numerical cases

Deployment Case

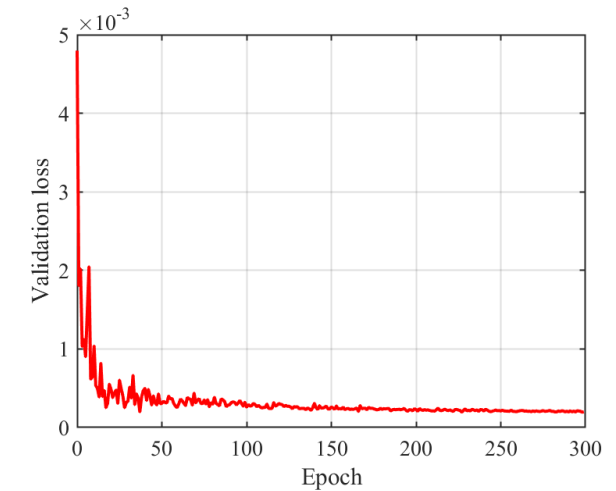
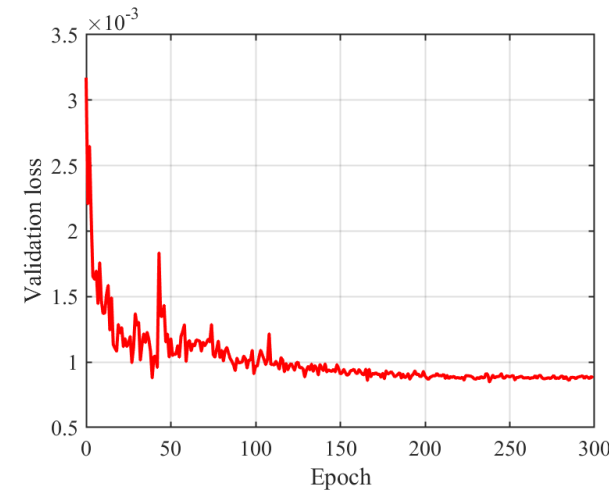
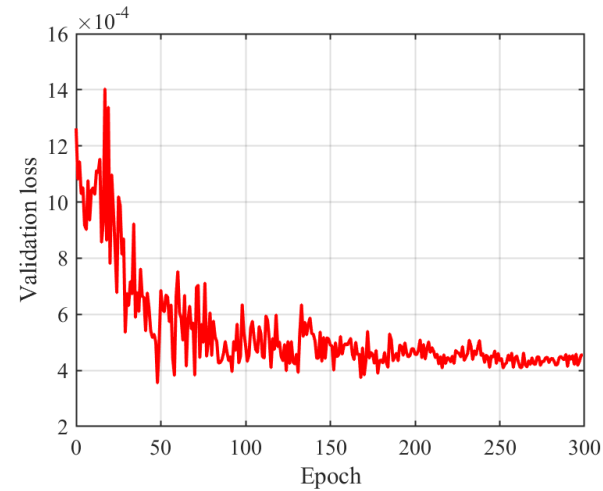
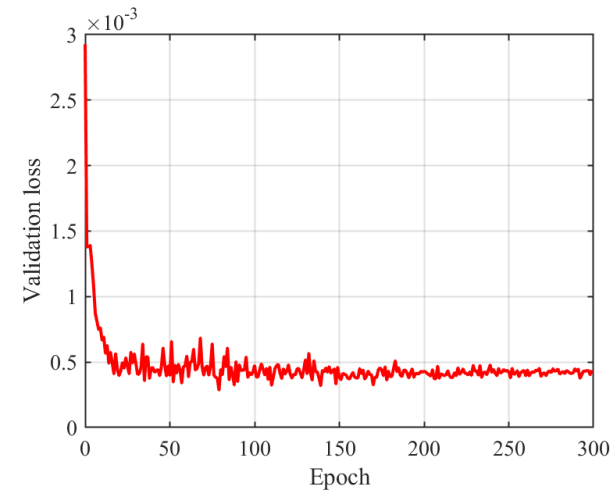
Initial state

$$\begin{aligned} \mathbf{x}_0 &= (\theta, \phi_1, \phi_2, l_1, l_2, \dot{\theta}, \dot{\phi}_1, \dot{\phi}_2, \dot{l}_1, \dot{l}_2)^T \\ &= (0, -30\text{deg}, -30\text{deg}, l_1, l_2, 0.02\text{ rad/s}, 0, 0, 0, 0)^T \end{aligned}$$

$$l_1 \in [2, 12] \text{ m}, l_2 \in [2, 12] \text{ m}$$

$$\begin{aligned} \mathbf{u}_0 &= (T_1, T_2, F_1, F_2)^T \\ &= (0.1, 0.1, 0, 0)^T \text{ N} \end{aligned}$$

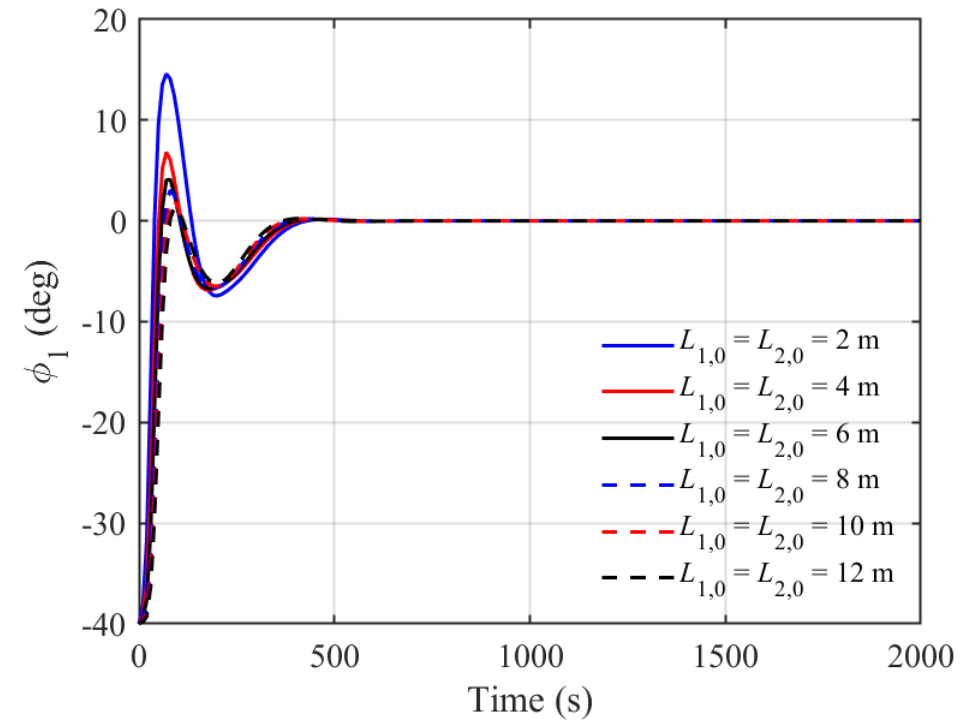
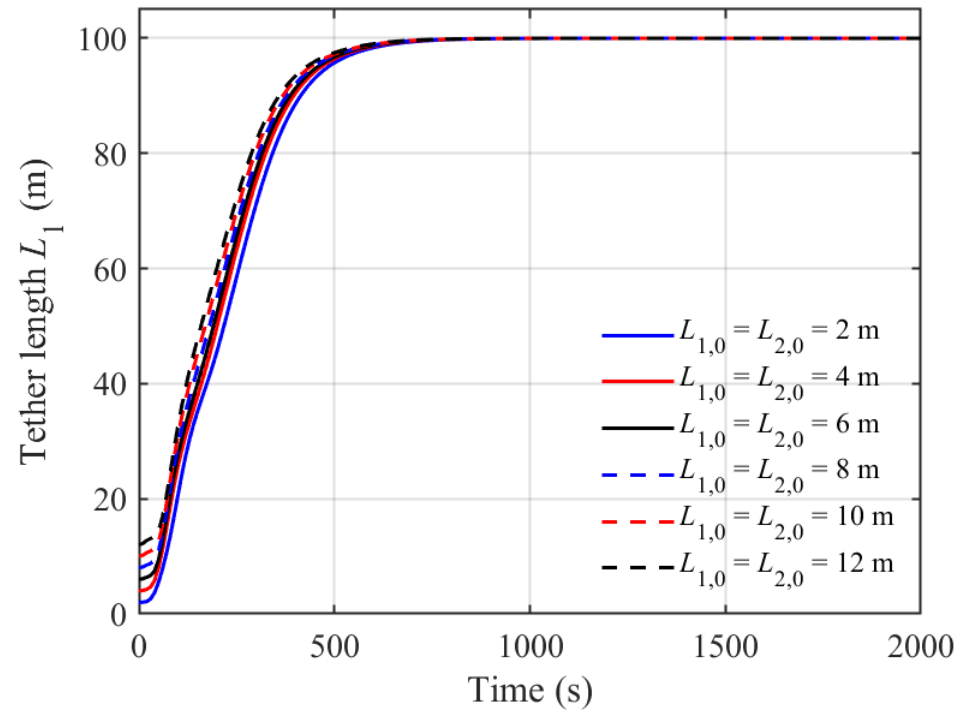
121 pairs of state-control data



Validation loss of the DNNs for $u_1 \dots u_4$

4 Numerical cases

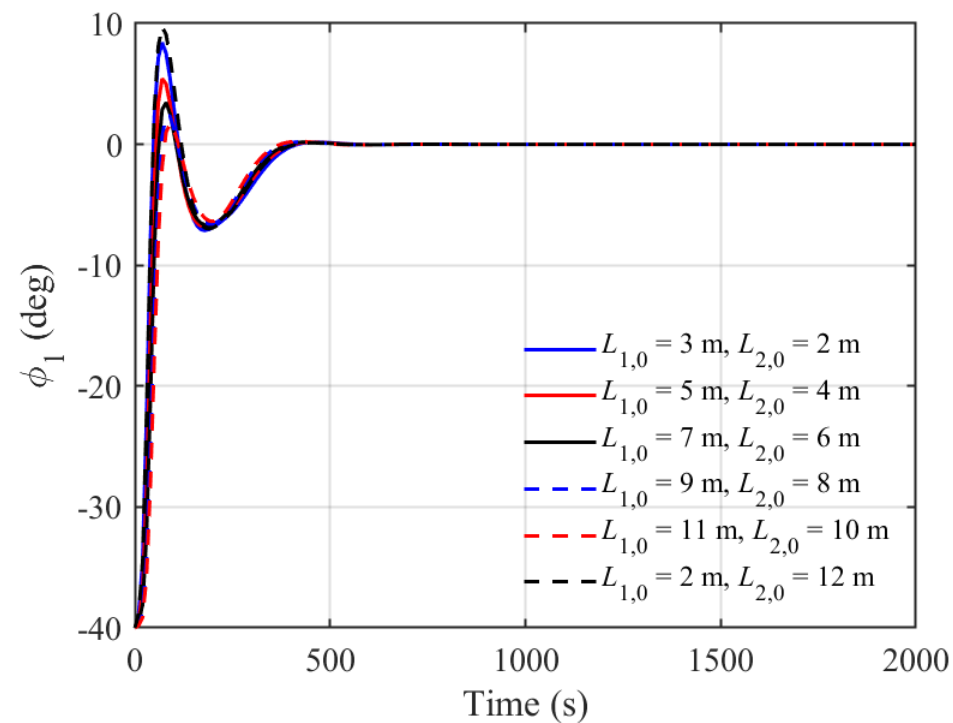
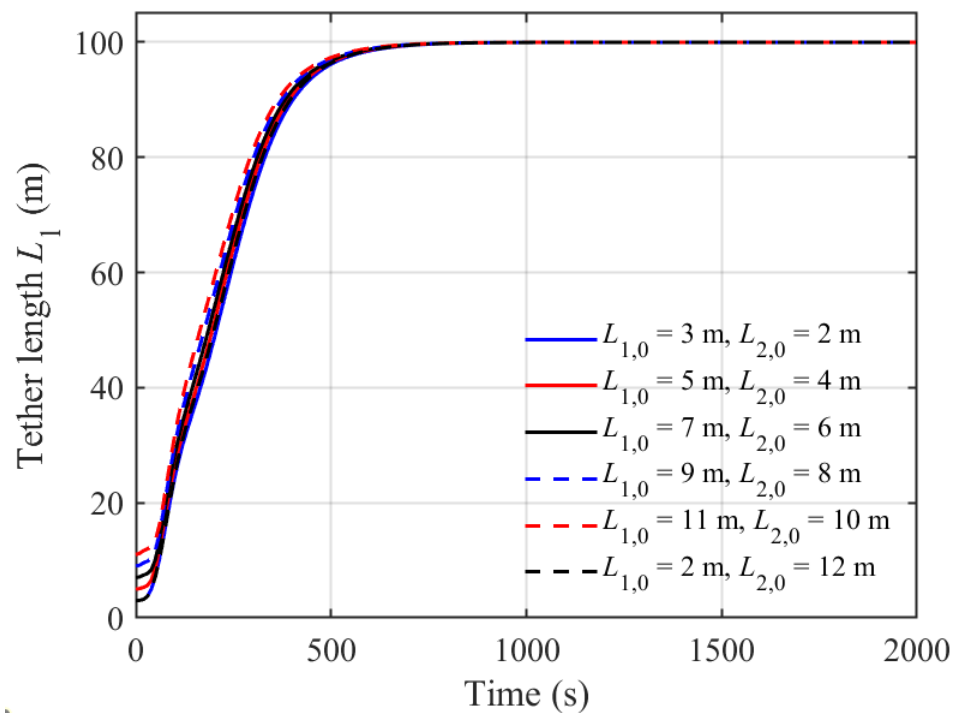
Deployment Case



Results of the system with initial **symmetric** configurations

4 Numerical cases

Deployment Case



Results of the system with initial **asymmetric** configurations

4 Numerical cases

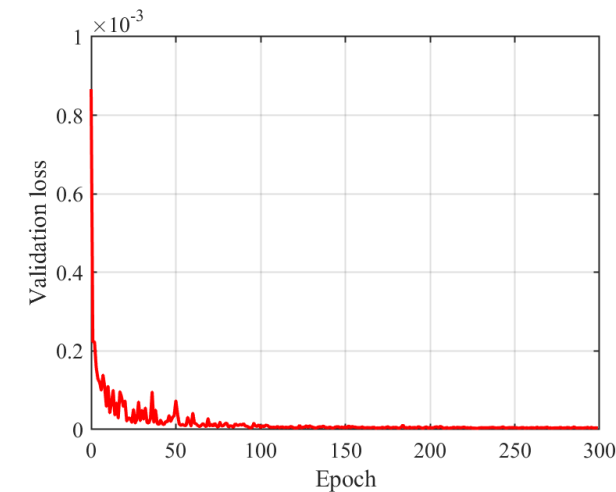
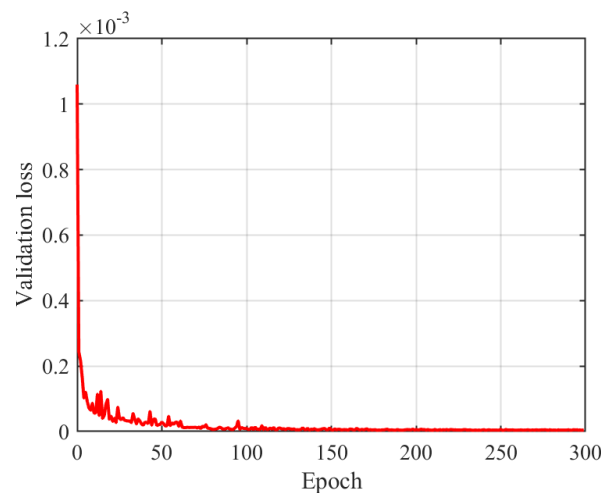
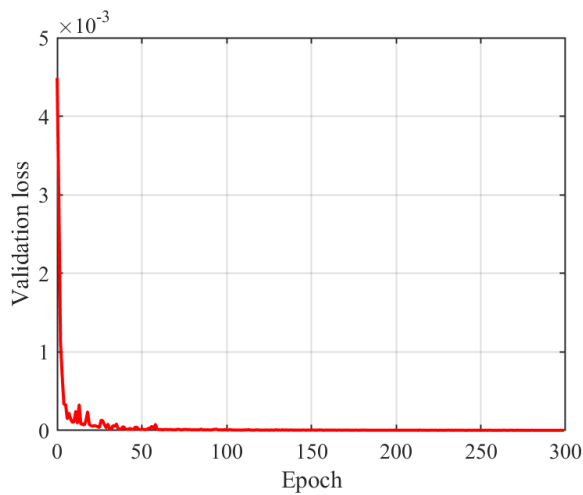
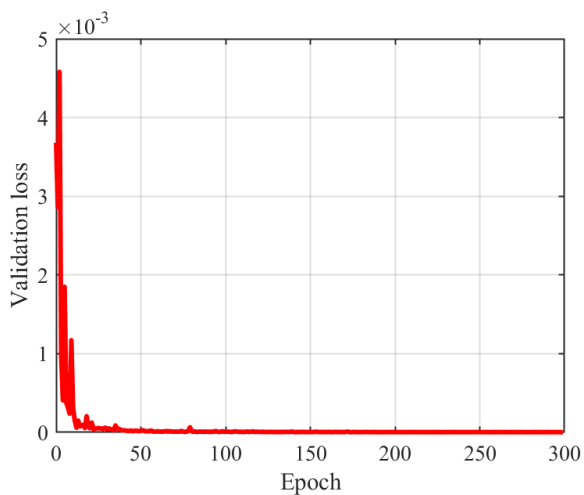
● Retrieval Case

➤ Initial state

$$\begin{aligned} \mathbf{x}_0 &= (\theta, \phi_1, \phi_2, l_1, l_2, \dot{\theta}, \dot{\phi}_1, \dot{\phi}_2, \dot{l}_1, \dot{l}_2)^T \\ &= (0, \phi_1, \phi_2, 100 \text{ m}, 100 \text{ m}, 0.02 \text{ rad/s}, 0, 0, 0, 0)^T \end{aligned}$$

$$\phi_1 \in [-30, -20] \text{ deg}, \quad \phi_2 \in [-30, -20] \text{ deg}$$

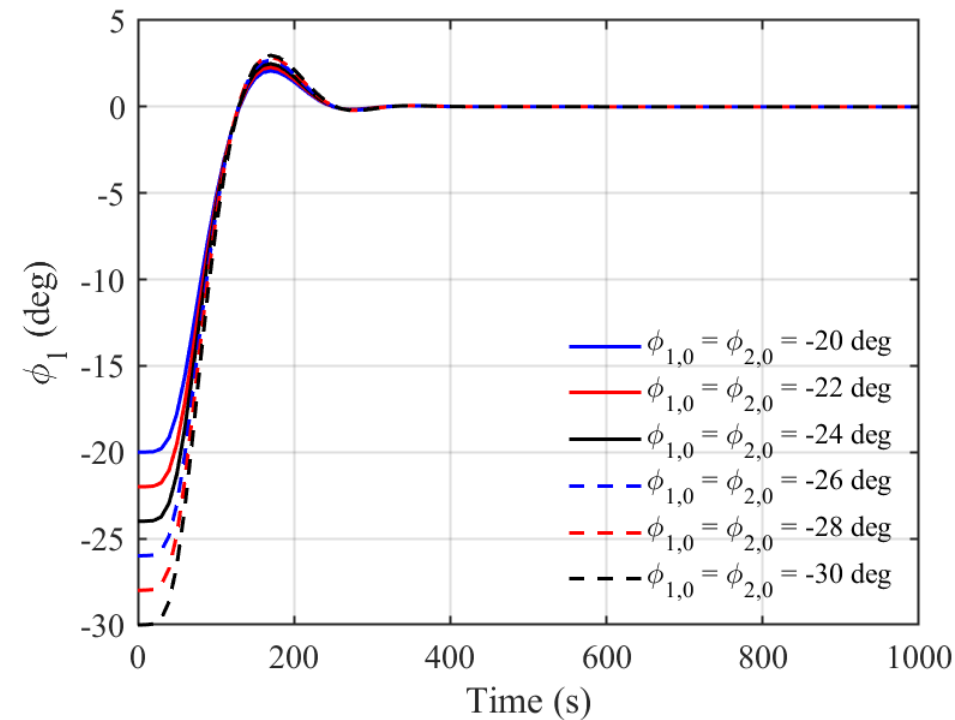
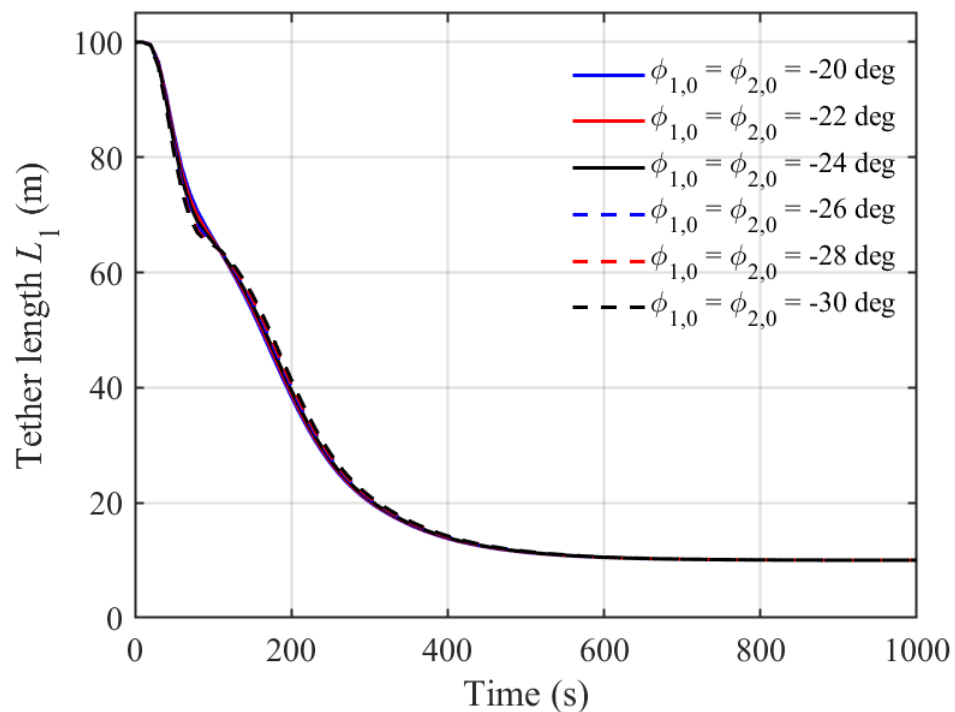
$$\begin{aligned} \mathbf{u}_0 &= (T_1, T_2, F_1, F_2)^T \\ &= (2.8, 2.8, 0, 0)^T \text{ N} \end{aligned}$$



Validation loss of the DNNs for u_1 --- u_4

4 Numerical cases

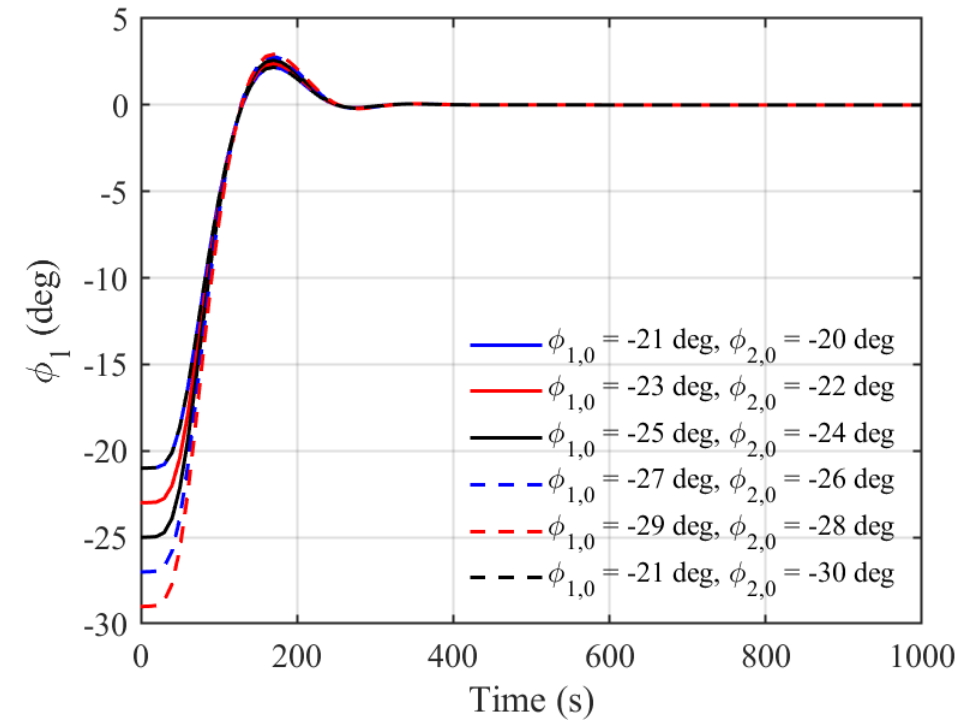
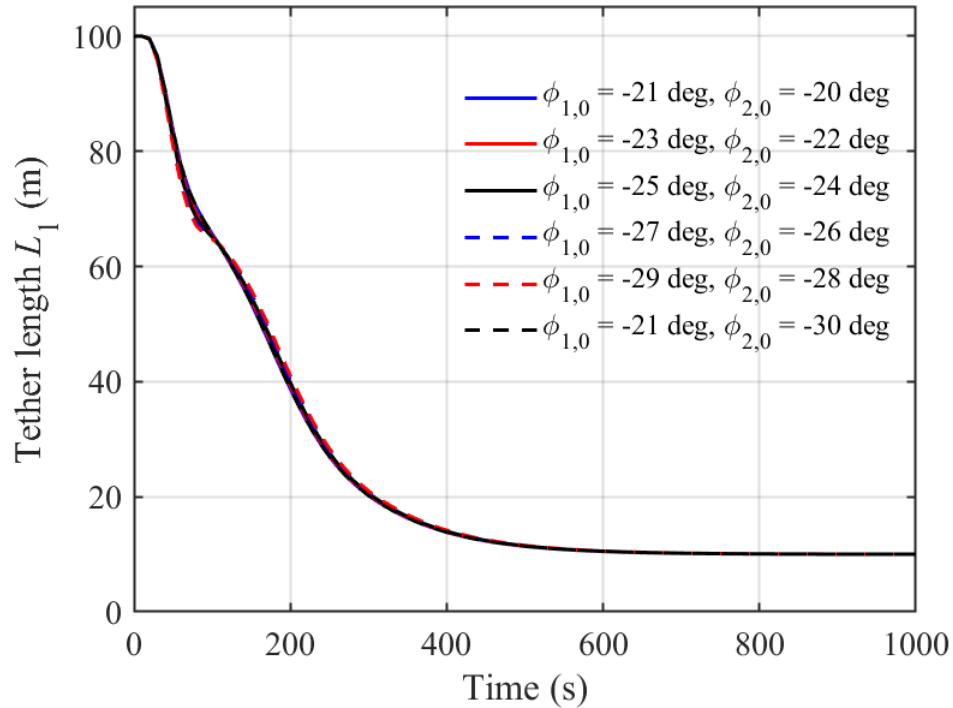
Retrieval Case



Results of the system with initial **symmetric** configurations

4 Numerical cases

Retrieval Case



Results of the system with initial **asymmetric** configurations

- The results show that the designed NMPC control law can well deal with the deployment and retrieval problems of the system with symmetric and asymmetric configurations and multiple constraints.
- The designed learning based controller can achieve the desired control goals using the DNNs obtained from the off-line dataset, which can greatly reduce the computational costs.
- In future work, a reinforcement learning-based controller is expected to be designed to achieve the quick deployment and retrieval control without the requirement of off-line pre-computed dataset.



Thank you!

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